



EXPLORING THE DEFLAGRATION-AUTOIGNITION-DETONATION TRANSITION IN THE CONTEXT OF PESSUE GAIN COMBUSTION

Authors: Roseline Ezekwesili¹, Camille Strozzi², Marc Bellenoue², Neda Djordjevic¹

¹ Technische Universität Berlin, Institute of Fluid Dynamics, 10623 Berlin, Germany ² Institut Pprime, CNRS, ISAE-ENSMA, Université de Poitiers, F-86962 Futuroscope Chasseneuil, France

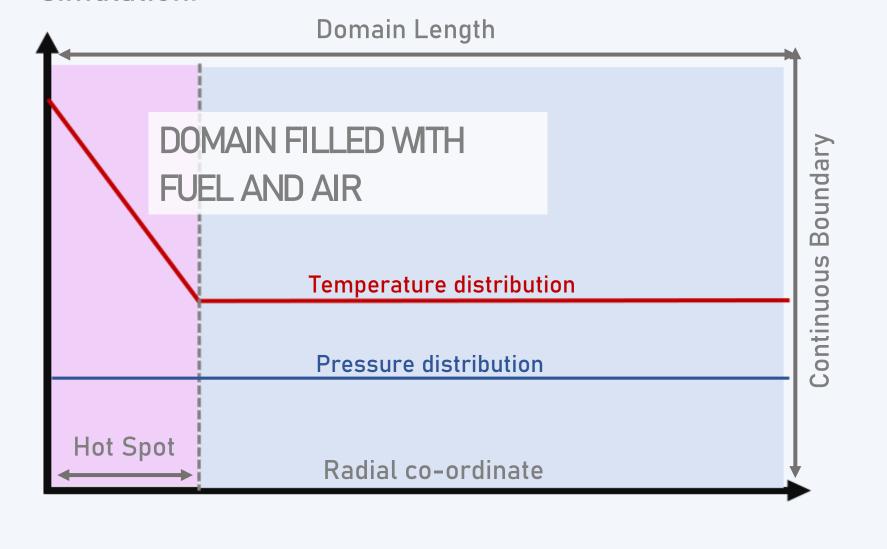
INTRODUCTION

PROJECT MOTIVATION REGIME DIAGRAM Autoignition of hot spots can develop into a detonation under The propagation modes of hot spots under different conditions can be represented by certain conditions. Detonations must be suppressed in Constant a regime diagram. The different hotspot conditions are represented by 2 Volume Combustion and promoted in Rotating Detonation dimensionless parameters. Combustion. Controlling the transition between combustion regimes is key to developing pressure gain combustion. Deflagration 12 PROJECT OBJECTIVES Subsonic reaction front u_a < a a = the local speed of sound STAGE 3 STAGE 1 radius r =**Detonation** Enhance Enhance the 1D model r_0 = hotspot radius understanding of to better match the $u_a = a$ t_e = excitation time autoignition experiment and t_{ia} = ignition delay time propagation explore strategies for u_a = autoignition wave speed modes using promoting and conventional 1D-T = temperature suppressing modelling detonations STAGE 2 $\varepsilon = (r_0/a)/t_e$ Supersonic Experimentally investigate the Deflagrationreaction front Autoignition-Detonation-Transition (DADT)

METHODOLOGY

04 STAGE 1 METHODOLOGY

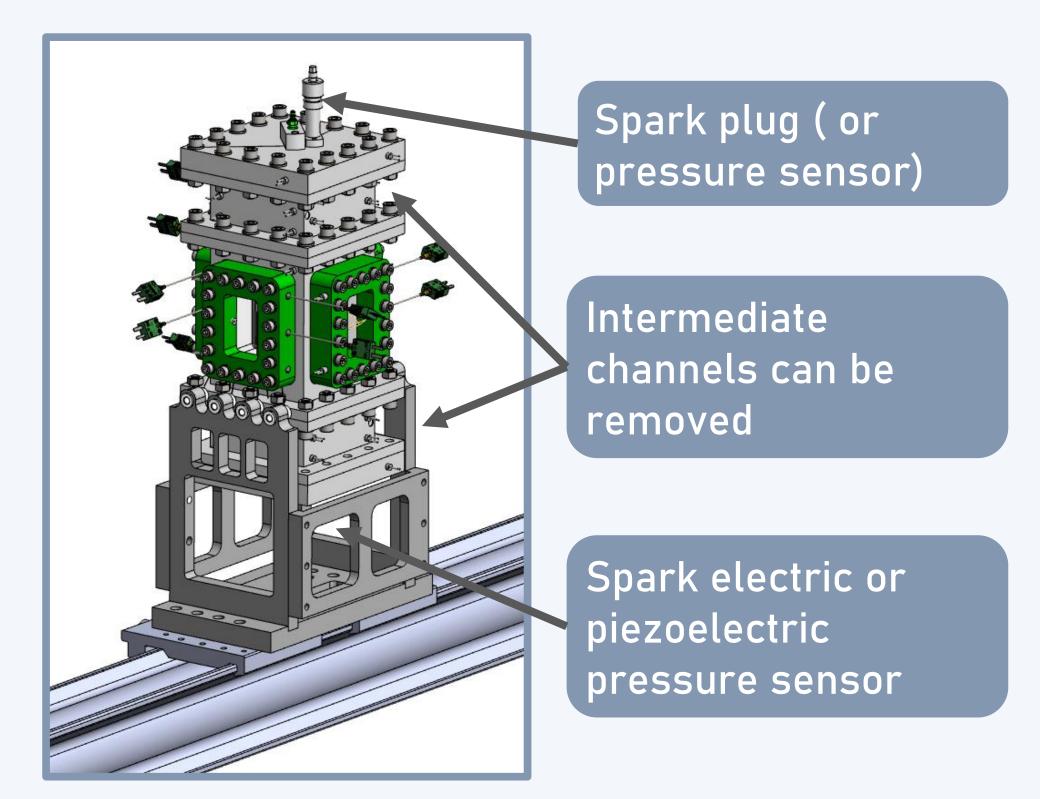
The INSFLA code provided by the Karlsruhe Institute of Technology is capable of providing time and space solutions for 1-D compressible reacting flows for both planar and spherical configurations. The diagram below shows the Initial conditions and set-up for a 1D spherical hotspot simulation.



Conditions Pressure, Temperature, Oxidizer
Explored Composition, Temperature Gradient

05 STAGE 2 METHODOLOGY

The DADT transition will be investigated experimentally using the MDAID set-up shown in the diagram below



❖ Modular square inner section 110/172/234 x 40 x 40 mm³

 $u_a > a$

- ❖ Spark plug with Transistorized Coil Igniter (E_{el}~30 mJ)
- Cartridge heaters with 1 to3 zones PID regulation
- Heat flux sensor (bottom)
- Four UV quartz windows: high speed chemiluminescence, schlieren, fluorescence...

STAGE 1 RESULTS

Figures 1, 2 and 3 are some of the results obtained from stage 1 using the INSFLA code. The results are for a stochiometric hydrogen and air mixture. The INSFLA code is still being adapted, therefore the results are only preliminary.

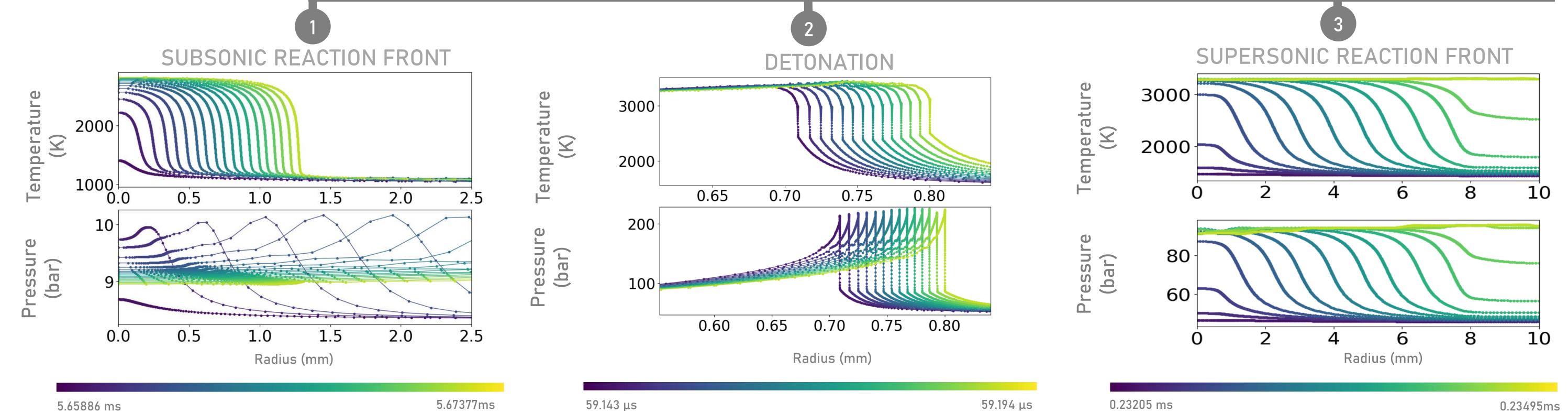


Figure 1: Stochiometric hydrogen in air at 1000K, 8 bar with a 2cm domain length and 2.5mm hotspot length and temperature gradient of 2K/mm in a planar configuration.

Figure 2: Stochiometric hydrogen in air at 1250K, 40 bar with a 10mm domain length, 8mm hotspot length and a temperature gradient of 0.125K/mm in a planar configuration.

Figure 3: Stochiometric hydrogen in air, at 1250K, 40 bar with a 10mm domain length, 8mm hotspot length and temperature gradient of 0.0125 K/mm in a planar configuration.